

Development of Fatigue Crack Growth Rates from Corner Crack Tests

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Luciano Smith, James Feiger, and Mark Thomsen
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Introduction

- Fatigue crack growth rates
 - As incorporated into AFGROW
 - As tested
 - ASTM E647
 - Current requirements
 - Upcoming changes
- Corner crack growth rate testing
 - Motivation
 - Description
 - Results

Growth rates in AFGROW

- Five options
 - Forman equation
 - Harter-T method
 - NASGRO (v3.0) equation
 - Tabular lookup
 - Walker equation
- Can develop smooth fit to data
- Tabular input gives most flexibility and accuracy with non-sigmoidal data

Tabular LOOKUP Data

Input values of Delta_K for da/dN values and up to 10 different R(stress ratio) values.
Matrix must have at least two R values and two da/dN values.
Input: Delta_K for R >= 0, input Kmax for R < 0.0

Number of da/dN Sets: 27 Number of R Sets: 2

		R[1]	R[2]
da/dN[1]	1.00e-009	2.606	1.38
da/dN[2]	3.00e-009	2.636	1.409
da/dN[3]	1.00e-008	2.673	1.503
da/dN[4]	2.00e-008	2.685	1.66
da/dN[5]	4.00e-008	2.720	1.807

Material name: User defined data

Ultimate Strength: 66 Young's Modulus: 10500

Coefficient of Thermal Expansion: 1.25e-005 Poisson's Ratio: 0.33

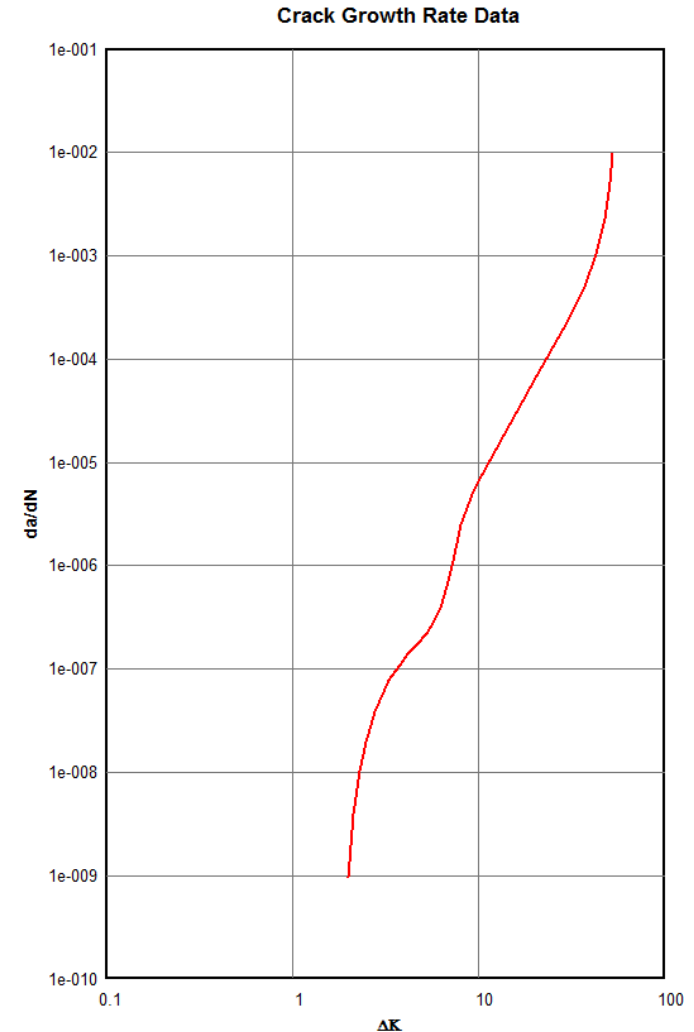
Upper limit on da/dN, DADNHI: 0.01 Lower limit on da/dN, DADNLI: 1e-009

Plane Stress Fracture Toughness, KIC: 62.777 Yield Strength, YLD: 47

Plane Strain Fracture Toughness, KIC: 35 Lower limit on R shift (Max. 0): -0.3

Delta K threshold value @R=0: 2.831 Upper limit on R shift (0, 1): 0.63

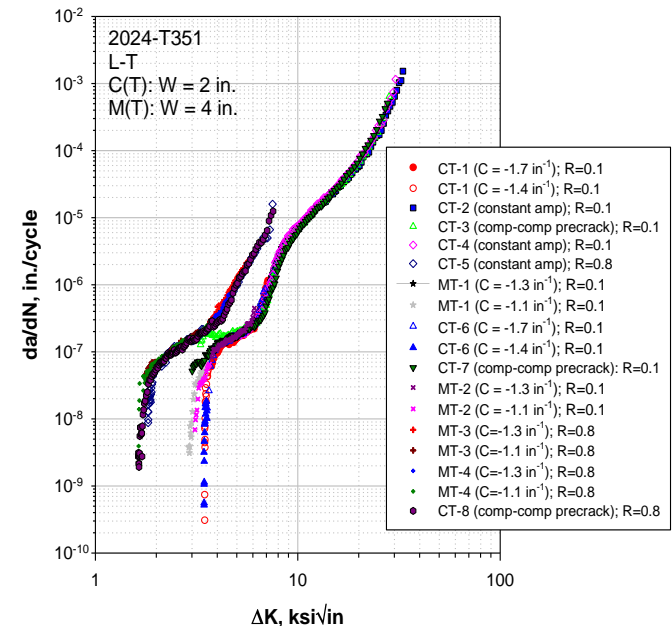
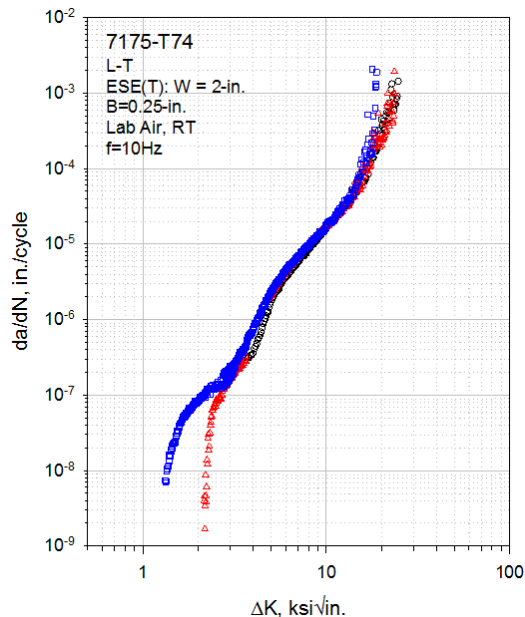
OK Cancel Save Read Apply



Note: For R < 0.0, Kmax is used instead of Delta K

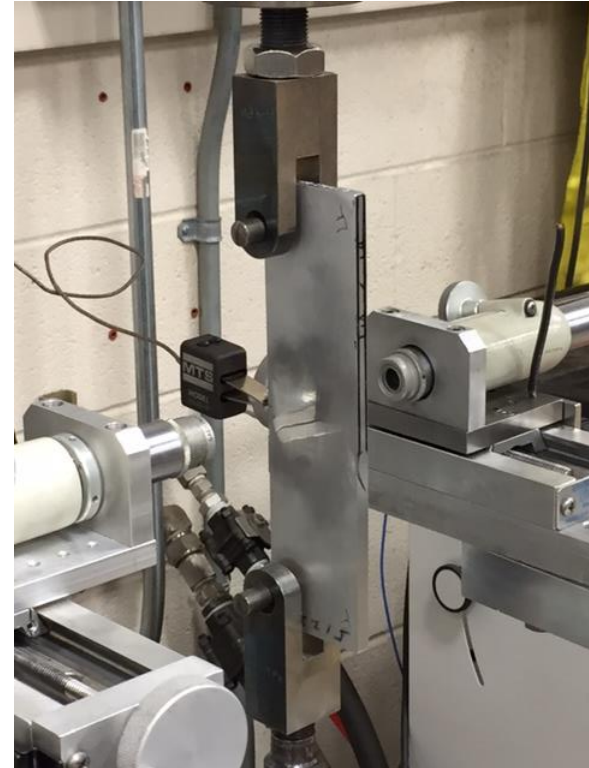
Growth rates as tested

- Test data can often be less smooth and therefore difficult to fit
- Variability when testing multiple specimens
 - Sometimes caused by residual stresses, crack branching, etc.
 - At lower growth rates and stress ratios, often crack closure
 - Plasticity induced – load history
 - Roughness induced – crack geometry and material microstructure



ASTM E647

- Standard Test Method for Measurement of Fatigue Crack Growth Rates
 - Specimen configuration
 - Test procedure
 - Calculation of growth rates
 - Reporting requirements



ASTM E647

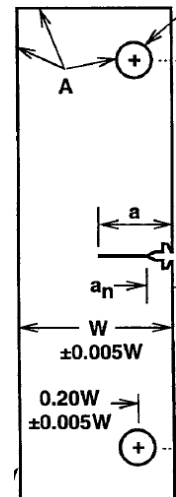
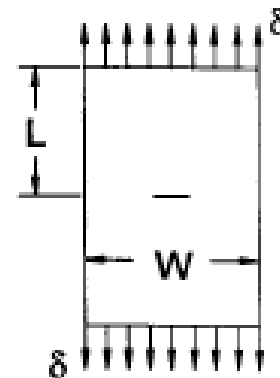
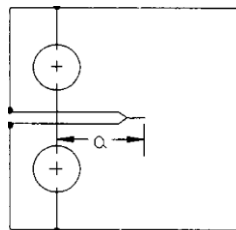
Standard Test Method for Measurement of Fatigue Crack Growth Rates

– Specimen configuration

- Any specimen type is allowed if the K solution is known
 - “Specimen configurations other than those contained in this method may be used provided that well-established stress-intensity factor calibrations are available”

- Three specimens are defined:

- Eccentrically-loaded single edge crack tension: ESE(T)
- Middle tension: M(T)
- Compact: C(T)



ASTM E647

■ Standard Test Method for Measurement of Fatigue Crack Growth Rates

- Specimen configuration

- Test procedure

 - Number of tests

 - Precracking method

 - Application of load

 - Constant force-amplitude or K-control for rates above 10^{-8} m/cycle

 - K-decreasing for rates below 10^{-8} m/cycle (near-threshold)

 - » Rate of load shedding is limited to prevent load history effects on growth rates (crack growth retardation)

$$C = \left(\frac{1}{K} \right) \cdot \left(\frac{dK}{da} \right) > -0.08 \text{ mm}^{-1} \left(-2 \text{ in.}^{-1} \right)$$

ASTM E647

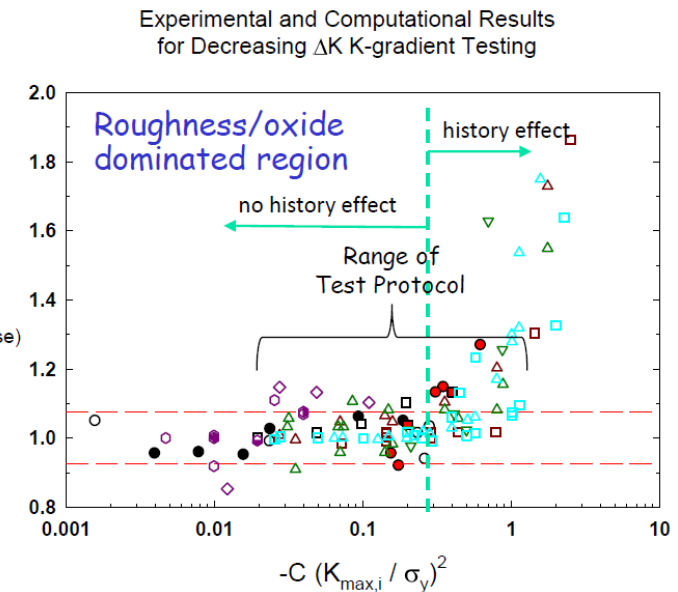
- Upcoming changes to near-threshold procedures
 - K-decreasing procedure being moved to an appendix
 - Consistency with compression precracking procedure, which is currently in an appendix
 - Load shedding gradient changing from C to a new unitless parameter:

$$-C \left(\frac{K_{max,i}}{\sigma_{YS}} \right)^2 < 0.1$$

$$\frac{\Delta K_{th}}{\Delta K_{th, baseline}}$$

or

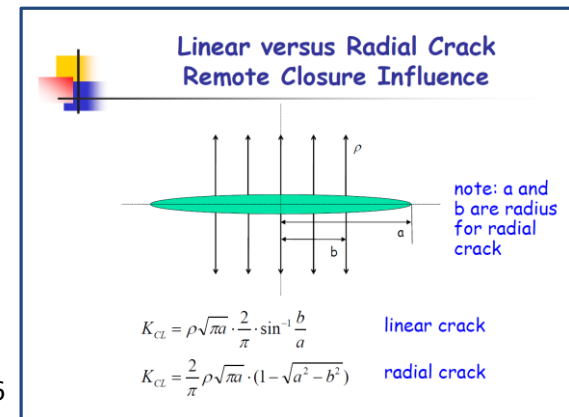
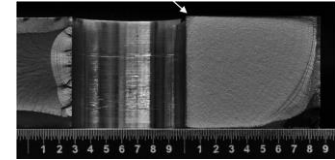
$$\frac{S_{op}/S_{max}}{S_{op}/S_{max}(base)}$$



(Ref: ASTM E08.06.06 meeting minutes, November 15, 2016)

Motivation for corner crack testing

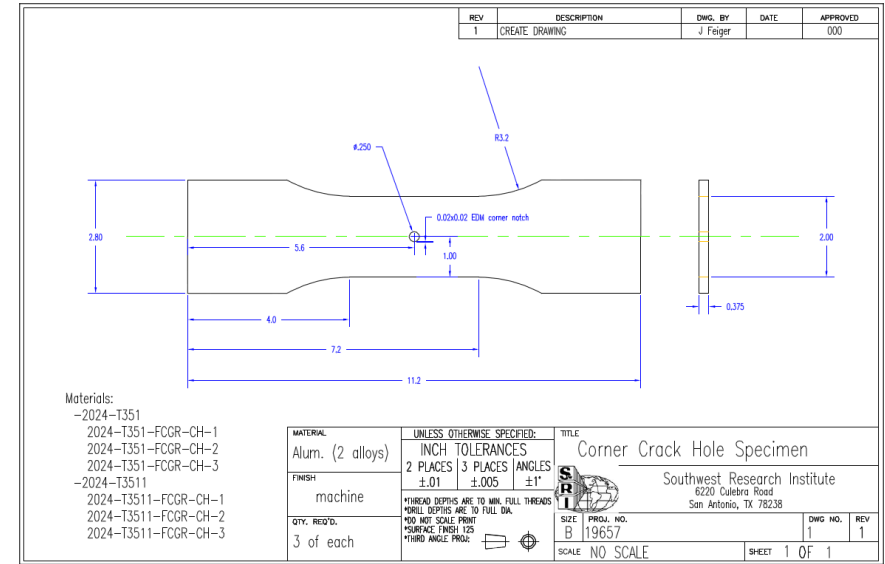
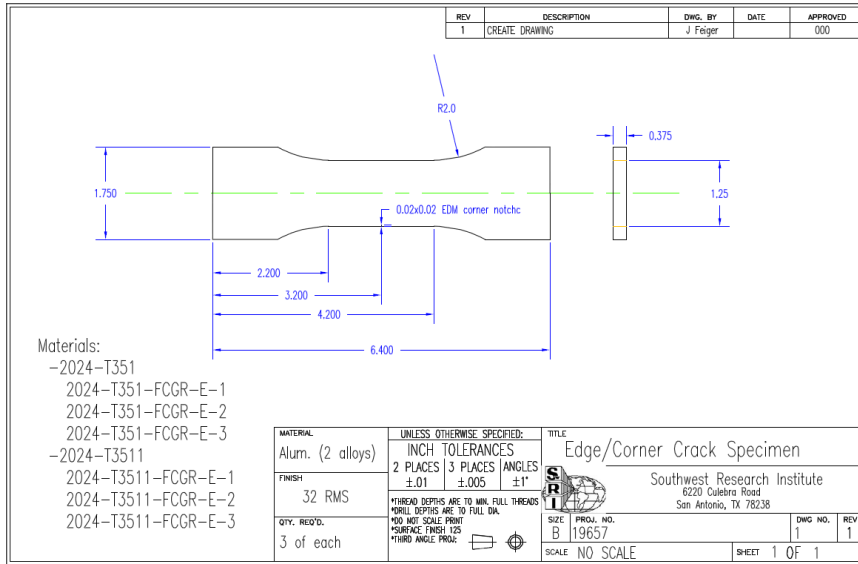
- Ability to gather L-T and L-S growth rate data in one test
- The standard specimens used for crack growth rate testing are all one-dimensional through cracks
 - The majority of analysis life is as corner crack
- When loading history is properly accounted for (minimizing plasticity induced crack closure), roughness induced closure dominates at low ΔK
 - Closure effect is smaller for radial crack versus linear crack (bulk material constraint)



(Ref: ASTM E08.06.06 meeting minutes, November 15, 2016)

Description of corner crack testing

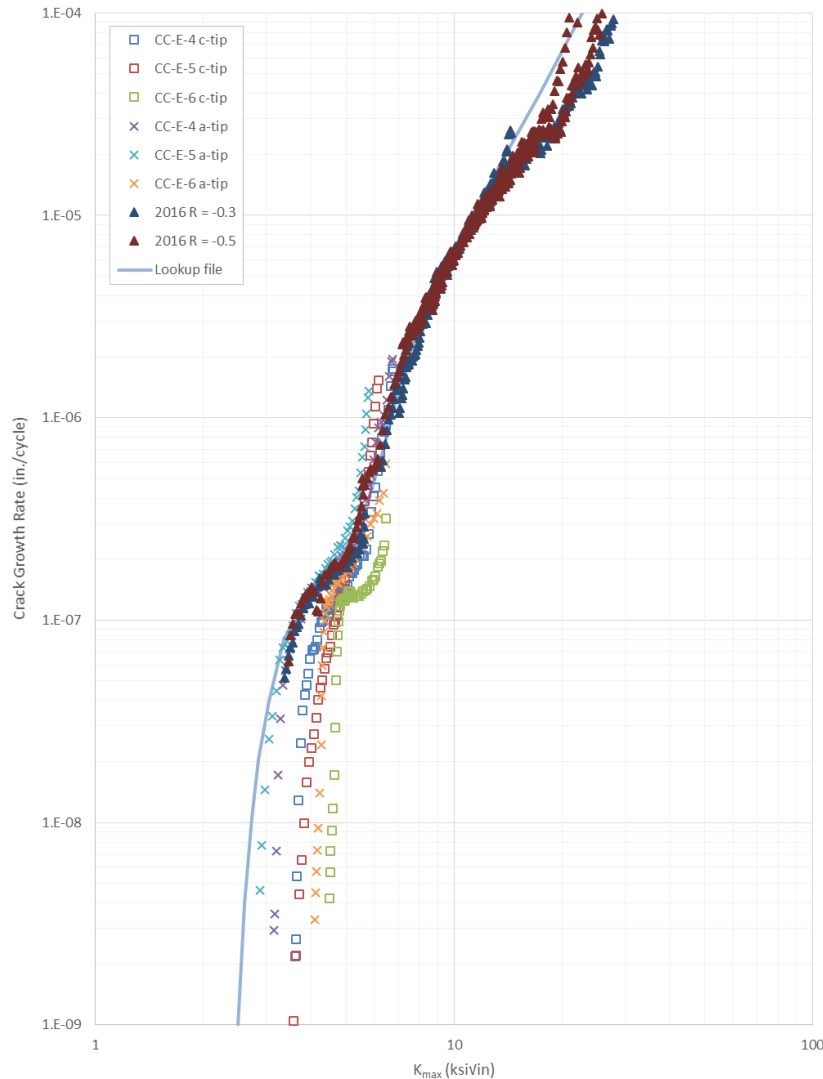
- All procedures follow E647, with two non-standard specimens



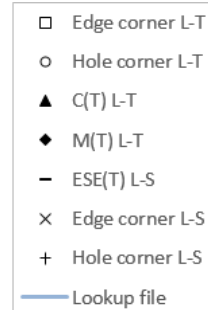
- R = 0.1 and -0.3
- Load shedding controlled by DCPD
 - $C = -4 \text{ in}^{-1}$ ($0.035 < -C (K_{\max,i} / \sigma_y)^2 < 0.097$)
 - Pre-test assumption of aspect ratios for a-tip K input
 - Post-test correction of applied K for da/dN-ΔK curves

Test results: T351 L-T and L-S, R = -0.3

2024-T351, L-T and L-S, R = -0.3 and -0.5

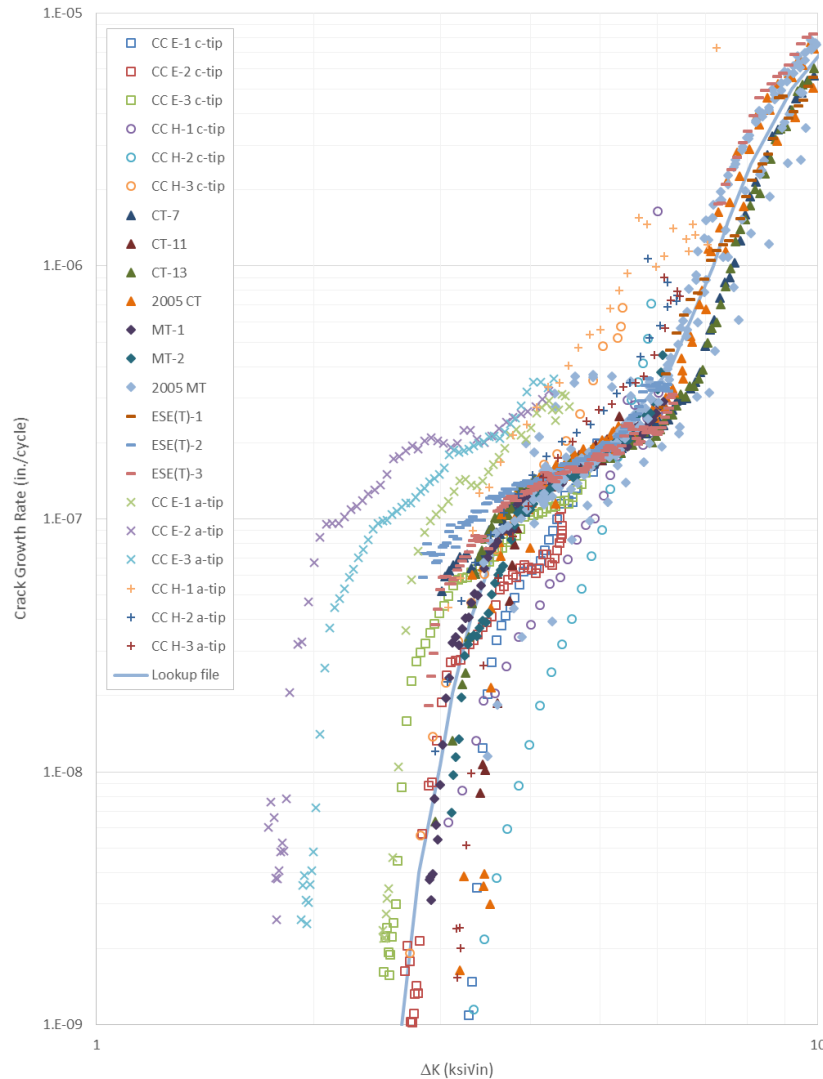


- Mostly consistent with M(T) data
- L-S (a-tip) data shows lower threshold than L-T (c-tip)
 - Very slightly lower than M(T)



Test results: T351 L-T and L-S, R = 0.1

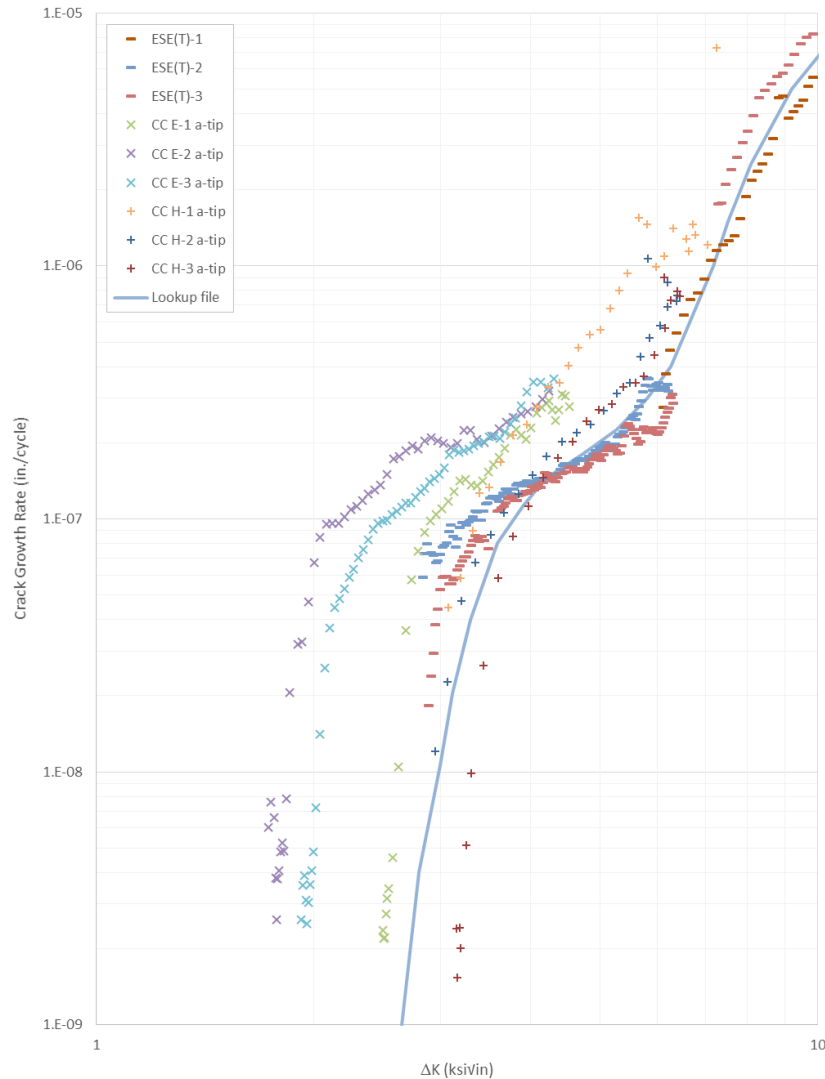
2024-T351, L-T and L-S, R = 0.1



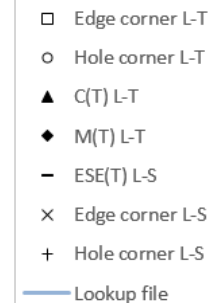
- L-S (a-tip and ESE(T)) data shows lower threshold than L-T (c-tip, C(T), and M(T)) data
- L-S data shows faster rates than the AFGROW lookup file
 - Potential for improved accuracy in corner crack aspect ratios

Test results: T351 L-S, R = 0.1

2024-T351, L-S, R = 0.1

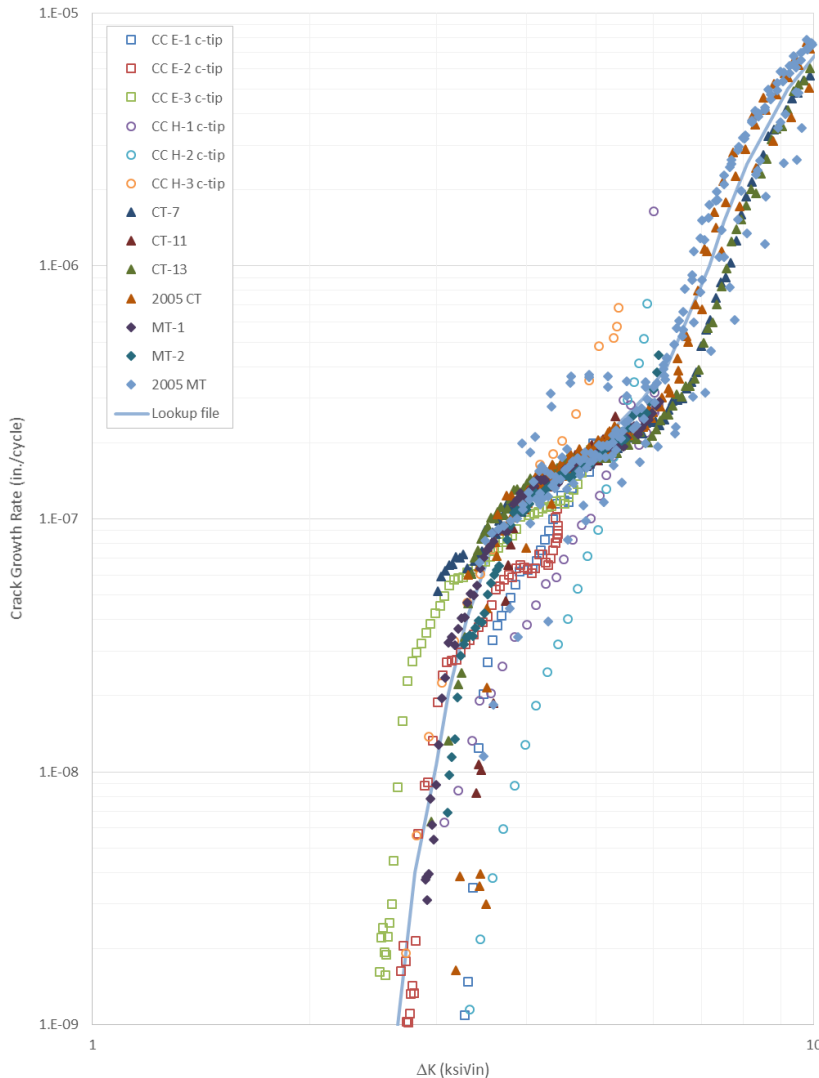


- Edge corner crack data shows lower threshold than both ESE(T) and hole corner crack



Test results: T351 L-T, R = 0.1

2024-T351, L-T, R = 0.1

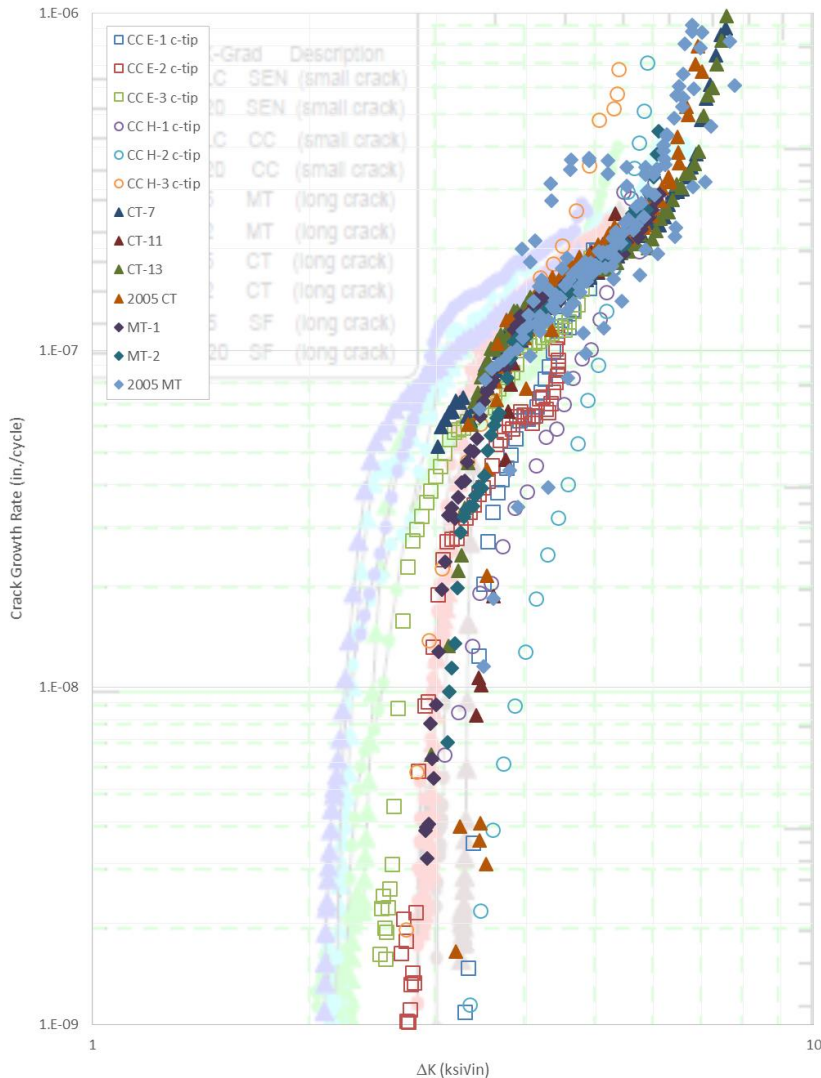


- Corner crack data consistent with C(T) and M(T) data
- Edge corner crack data shows lower threshold than hole corner crack



Test results: T351 L-T, R = 0.1

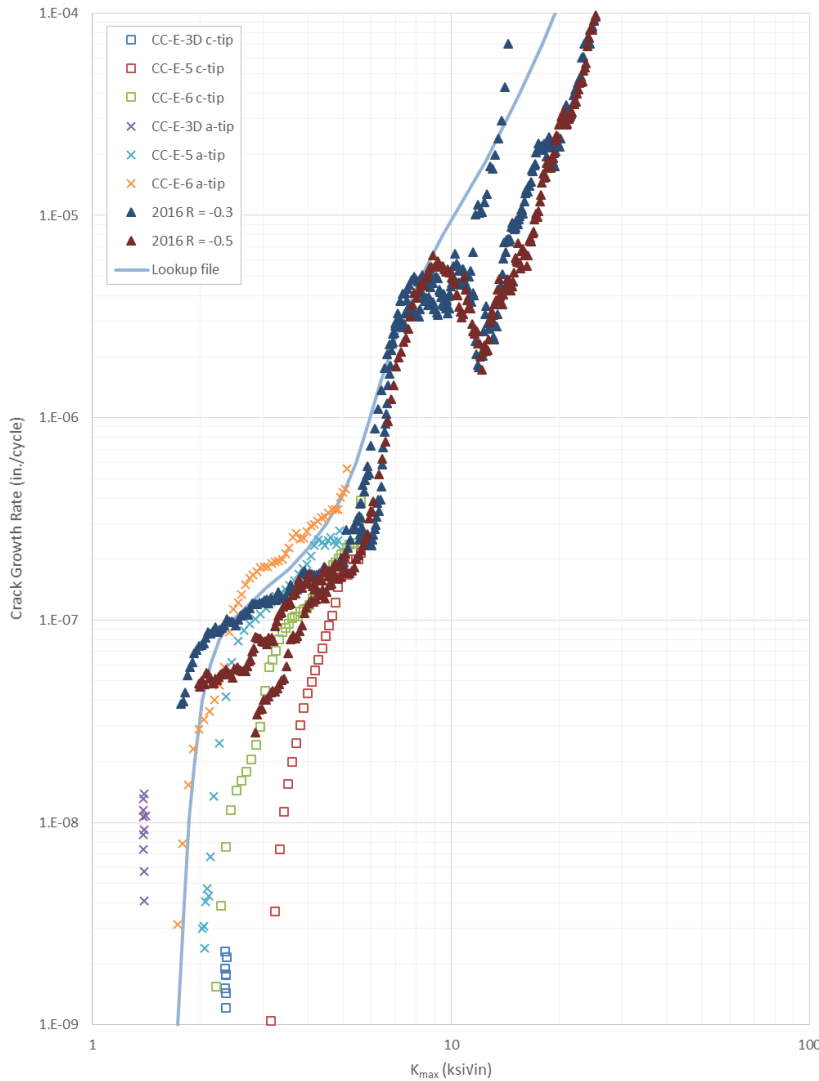
2024-T351, L-T, R = 0.1



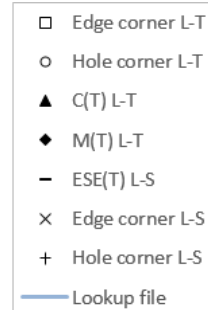
- Comparison against Keith Donald data presented at ASTM E08.06, November 2016
 - Through crack results match very well
 - Corner crack results:
 - SwRI consistent with through crack results
 - Donald shows smaller threshold
- Inconsistency caused by C?
 - SwRI: -4 in^{-1}
 - Donald: -20 in^{-1}
 - Δa range $\sim 0.04''$

Test results: T35 | | L-T and L-S, R = -0.3

2024-T3511, L-T and L-S, R = -0.3 and -0.5

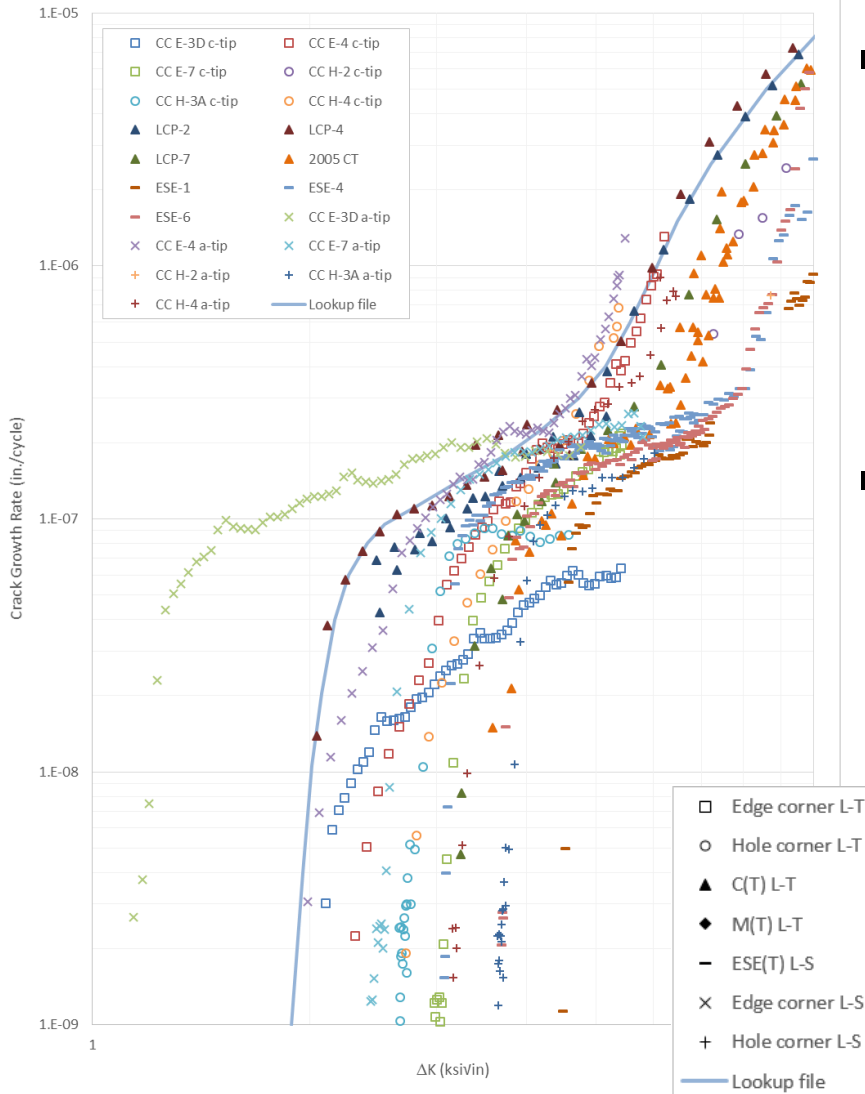


- Mostly consistent with M(T) data
- L-S (a-tip) data shows lower threshold than L-T (c-tip)



Test results: T35 | | L-T and L-S, R = 0.1

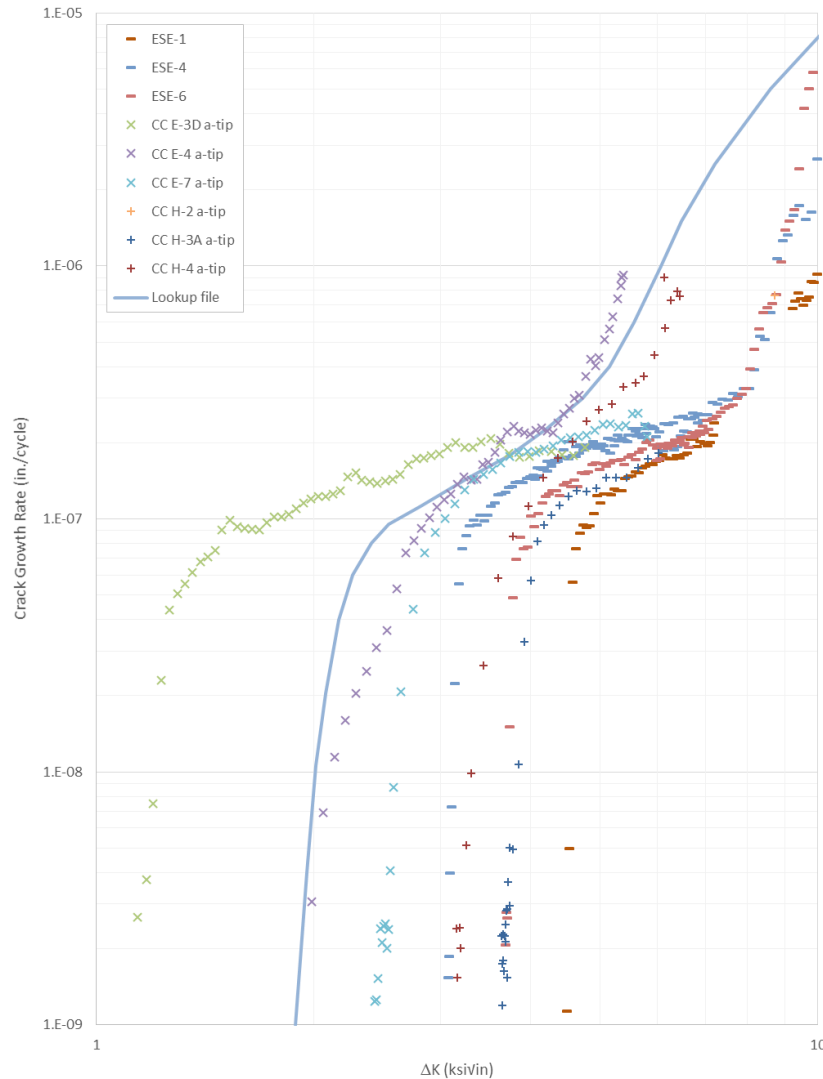
2024-T3511, L-T and L-S, R = 0.1



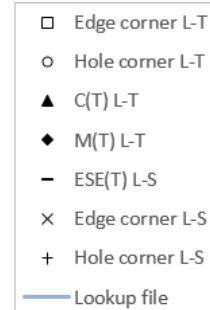
- L-S (a-tip and ESE(T)) and L-T (c-tip, C(T), and M(T)) data show similar threshold values
 - Not including one outlier
- Corner crack and through crack data show lower rates than the AFGROW lookup file
 - Lookup file is conservative, but not unrealistic
 - Not including one outlier

Test results: T35 | | L-S, R = 0.1

2024-T3511, L-S, R = 0.1

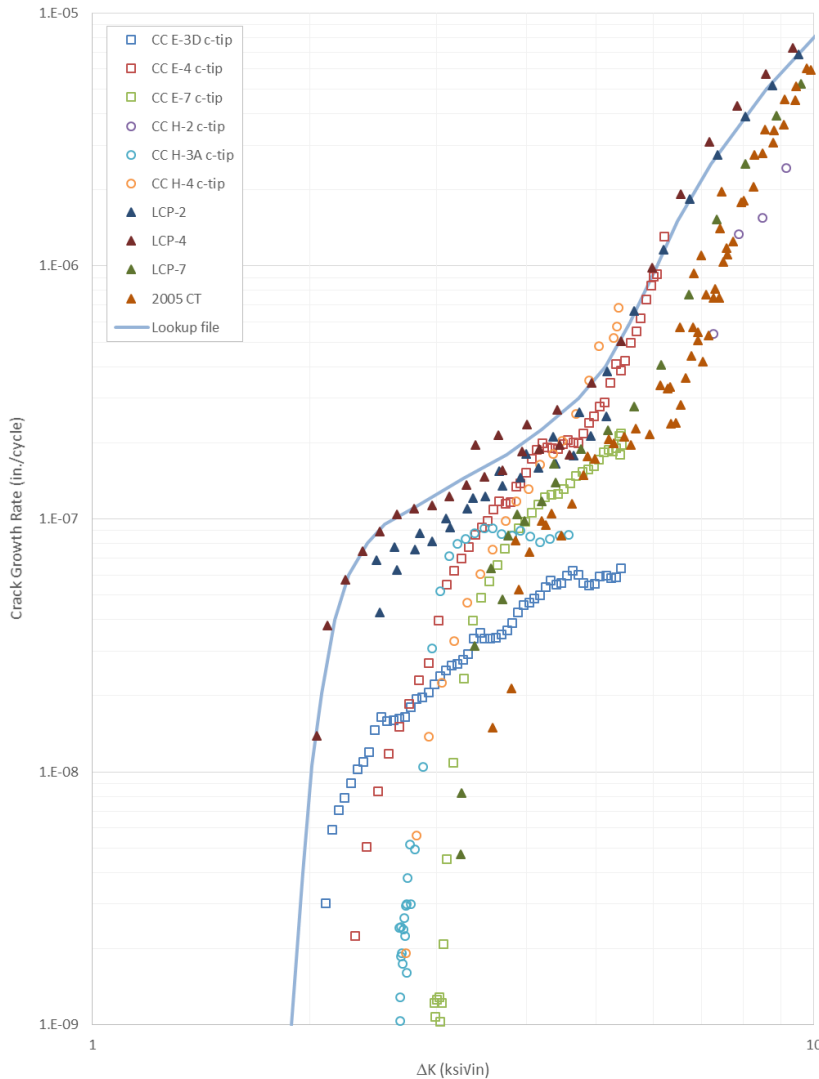


- Edge corner crack data shows lower threshold than both ESE(T) and hole corner crack

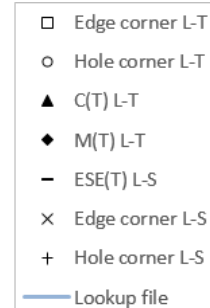


Test results: T35 | | L-T, R = 0.1

2024-T3511, L-T, R = 0.1



- Corner crack data consistent with C(T) and M(T) data
- Edge and hole corner crack rates are similar



Conclusions

- Successfully developed $da/dN-\Delta K$ curves from E647 testing using corner crack specimens
- Data developed for both L-T and L-S cracking
 - Simpler method for L-S data than using through crack specimens
 - Thin specimens possible
- Method did not decrease variability seen in near-threshold data
 - Cracked edge specimens more consistent and more in line with expectations than cracked hole specimens

