

Northrop – Grumman Workshop 2012

AFGROW

Alex Litvinov, James Harter,
LexTech, Inc

Overview

- About AGROW
- New features in the upcoming AFGROW release 5.2
- New features in release 5.3
- Q&A

About AFGROW

AFGROW is a Damage Tolerance Analysis (DTA) framework that allows users to analyze crack initiation, fatigue crack growth, and fracture to predict the life of metallic structures.

<http://www.afgrow.net>

<http://www.facebook.com/afgrow>



Customers

Advanced Technologies Incorporated
Air Force Research Laboratory
Air Spray Aviation Services
Altep
Australian Aerospace
Australian Department of Defense
Bell Helicopter
Caterpillar Inc.
Cessna
Circor Aerospace
Create
DART Aerospace
Delta Engineering
European Science Foundation
Gulfstream
Italian Air Force
Instytut Techniczny Wojsk Lotniczych /ITWL
ITT
Institute for Materials Applications in Mechanical Engineering
Japan Aerospace Exploration Agency
Kalitta Air, LLC
L-3
Live TV
Lockheed Martin
Marshall Aerospace

Micro Craft, Inc.
MIT
Mitre
NASA
Northrop Grumman
NRC Institute for Aerospace Research
NTSB
Ogden Air Logistics Center
PACS Engineering
P.B. Engineering Inc.
Pilatus Aircraft Ltd
Precision Engineering LLC
Raytheon
TanBeräkning AB
Tenaris
Tcagi
Textron India
Triumph Group, Inc
TUI Airlines
United States Air Force Academy
University of Ulsan
The University of Utah
Virginia Tech
VZLU - Aeronautical Research & Test Institute
Western Avionics
Wyle

New Release

- Extended Advanced Model Solution for Corner Cracks at a Hole to Handle Small Cracks
- Updated Solution for a Single Through Crack at a Hole Under Bearing Load
- Improved Offset Correction for Cracks at Holes Under Bearing Load
- Multi-Channel Spectrum Format
- Spectrum Tool
- Warning Messages When K-Solution Limits are Exceeded
- ~~• Ability to Replicate Results From Previous Versions (back to Version 5.01)~~
- Expanded Input Table Size (virtually unlimited)
- ~~• Option to Save Input File With Retardation State Data for Later Restart~~
- Improved COM and Plug-In Interface
- New GUI
- Corner Crack at a Countersunk Hole Solution
- Beta Sub-R

NEW GUI

The screenshot displays the AFGROW software interface with the following components:

- Menu Bar:** File, Input, Edit, View, Predict, Tools, Repair, Initiation, Window, Help
- Status Panel (Left):**
 - Example Problem
 - Specimen
 - 2024 T-3 Bare Sheet LONG CRACK DATA (Harter T-method)
 - Stress State
 - Spectrum
 - No Spectrum Filters
 - No Retardation
 - No K-Solution Filters
 - No Residual Stresses
- Specimen View (Center):** A diagram of a specimen with a central crack and loading points.
- Properties Panel (Right):**
 - Specimen
 - Appearance: (Name) Specimen
 - Size: Width 4.000000, Thickness 0.250000
 - Load: Axial 1.000000, Bending 0.000000, Bearing 0.000000
 - Solution: -
 - Remove All
- Output Window (Bottom):**

```

Crack #1
Left Tip C= 0.11072 Beta Tension=1.4760 Beta Compression=1.4760 R(k)= 0.0000 R(final)= 0.0000
Right Tip C= 0.11072 Beta Tension=1.4760 Beta Compression=1.4760 R(k)= 0.0000 R(final)= 0.0000
Left Tip A= 0.22901 Beta Tension=1.0956 Beta Compression=1.0956 R(k)= 0.0000 R(final)= 0.0000
Right Tip A= 0.22901 Beta Tension=1.0956 Beta Compression=1.0956 R(k)= 0.0000 R(final)= 0.0000
Max stress 14.000, r = 0.00, 21600 Cycles, Constant amp.: 217, Pass: 217
Delta k=1.2188e+001 D()/DN=5.8241e-006
Delta k=1.2188e+001 D()/DN=5.8241e-006
Delta k=1.3010e+001 D()/DN=7.0812e-006
Delta k=1.3010e+001 D()/DN=7.0812e-006

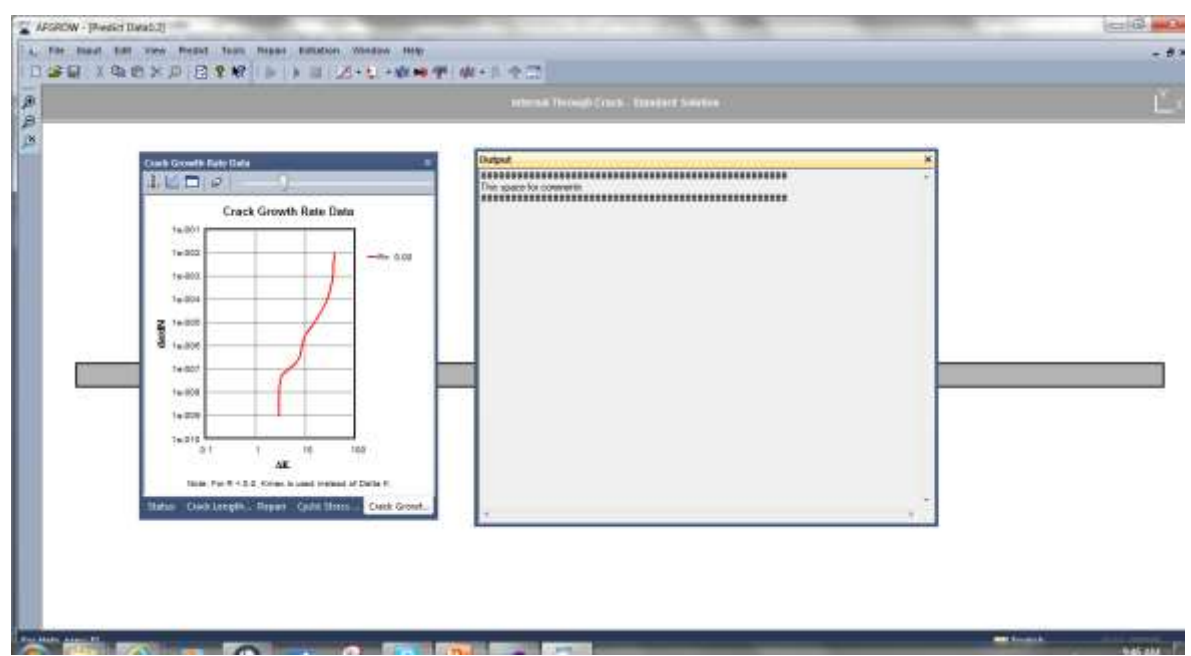
Crack #1
Left Tip C= 0.1113 Beta Tension=1.4760 Beta Compression=1.4760 R(k)= 0.0000 R(final)= 0.0000
Right Tip C= 0.1113 Beta Tension=1.4760 Beta Compression=1.4760 R(k)= 0.0000 R(final)= 0.0000
Left Tip A= 0.22371 Beta Tension=1.0956 Beta Compression=1.0956 R(k)= 0.0000 R(final)= 0.0000
Right Tip A= 0.22371 Beta Tension=1.0956 Beta Compression=1.0956 R(k)= 0.0000 R(final)= 0.0000
Max stress 14.000, r = 0.00, 21700 Cycles, Constant amp.: 218, Pass: 218
Delta k=1.2220e+001 D()/DN=5.8700e-006
Delta k=1.2220e+001 D()/DN=5.8700e-006
Delta k=1.3030e+001 D()/DN=7.1140e-006
Delta k=1.3030e+001 D()/DN=7.1140e-006

```

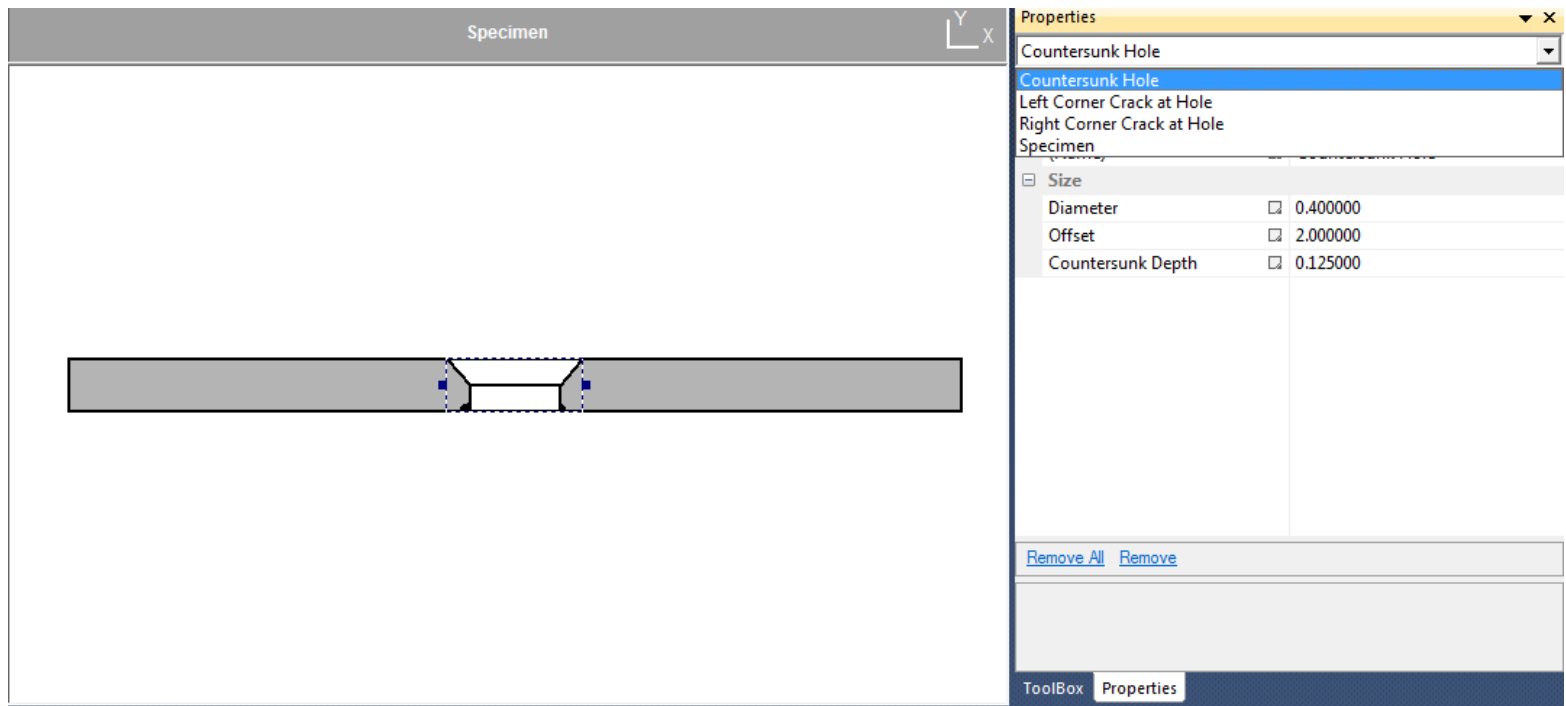
Output Frame – Copy and Paste



Floating Dockable Views



Advanced Model – Switch Objects

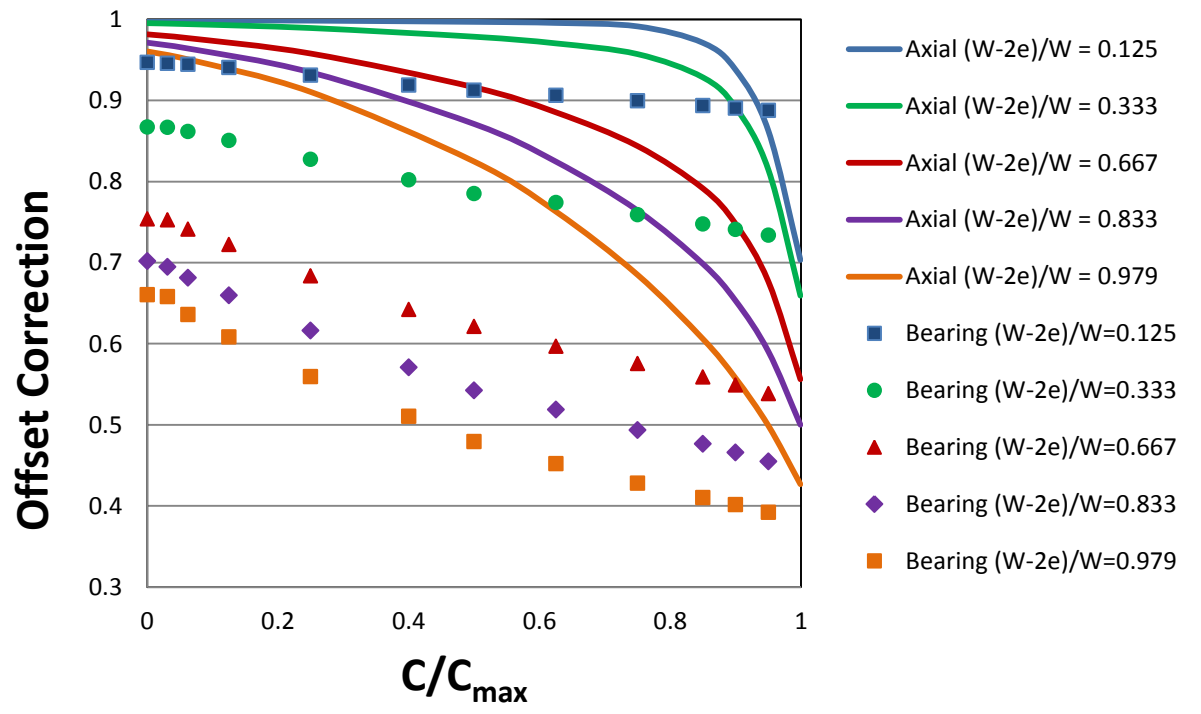
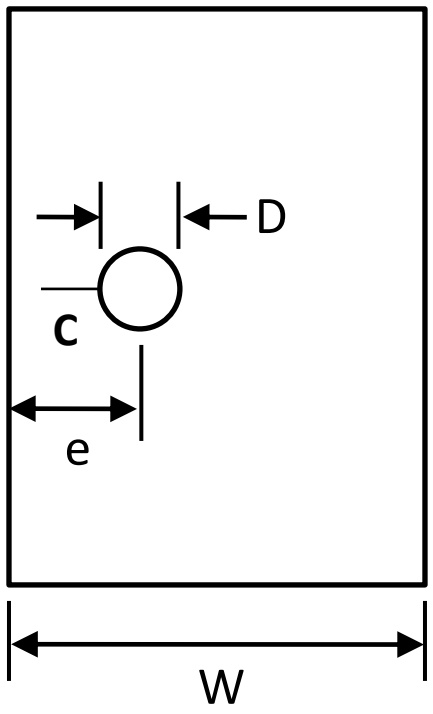


The screenshot displays a CAD application window titled "Specimen". The main view shows a 3D model of a cylindrical specimen with a countersunk hole. The hole is highlighted with a dashed blue outline, and small blue squares indicate the hole's position and dimensions. To the right, a "Properties" panel is open, showing a list of objects: "Countersunk Hole", "Left Corner Crack at Hole", "Right Corner Crack at Hole", and "Specimen". The "Countersunk Hole" object is selected, and its properties are displayed in a table below:

Size	
Diameter	<input checked="" type="checkbox"/> 0.400000
Offset	<input checked="" type="checkbox"/> 2.000000
Countersunk Depth	<input checked="" type="checkbox"/> 0.125000

At the bottom of the properties panel, there are two buttons: "Remove All" and "Remove". The bottom status bar of the application shows "ToolBox" and "Properties" tabs.

New Offset Hole Solution Bearing Load Case



Solution Matrix

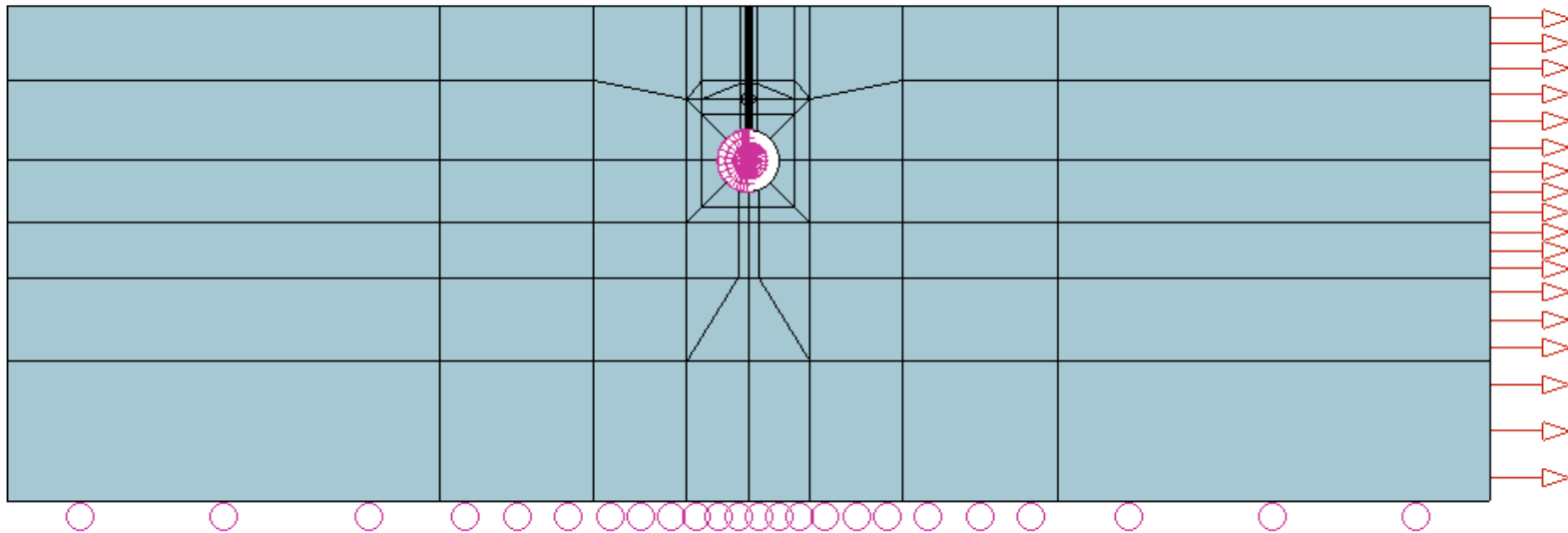
e/D : 0.75, 1, 2, 4, 8, 16

$(W-2e)/W$: 0.125, 0.25, 0.333, 0.667, 0.833,
0.918, 0.958, 0.979

C/C_{\max} : 0 to 0.95

Note: The zero crack length correction was determined by the ratio of the local K_T values

FEM* Boundary Conditions



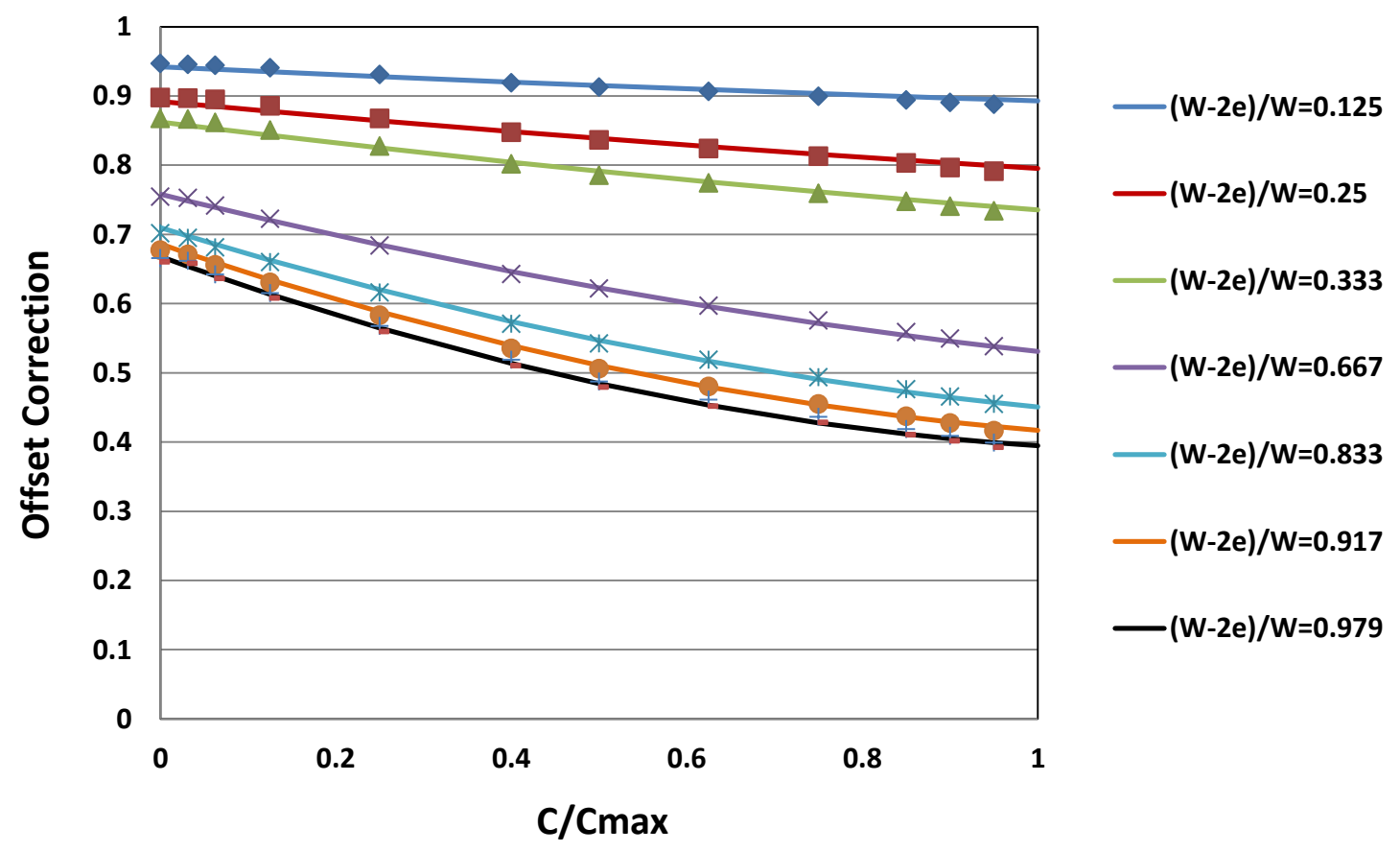
Model Thickness = 1.0

Loading to produce a unit resultant force at the hole ($1/W$)

Spring ($E_{\text{Spring}} = 3X E_{\text{Plate}}$) B.C. along $\frac{1}{2}$ hole

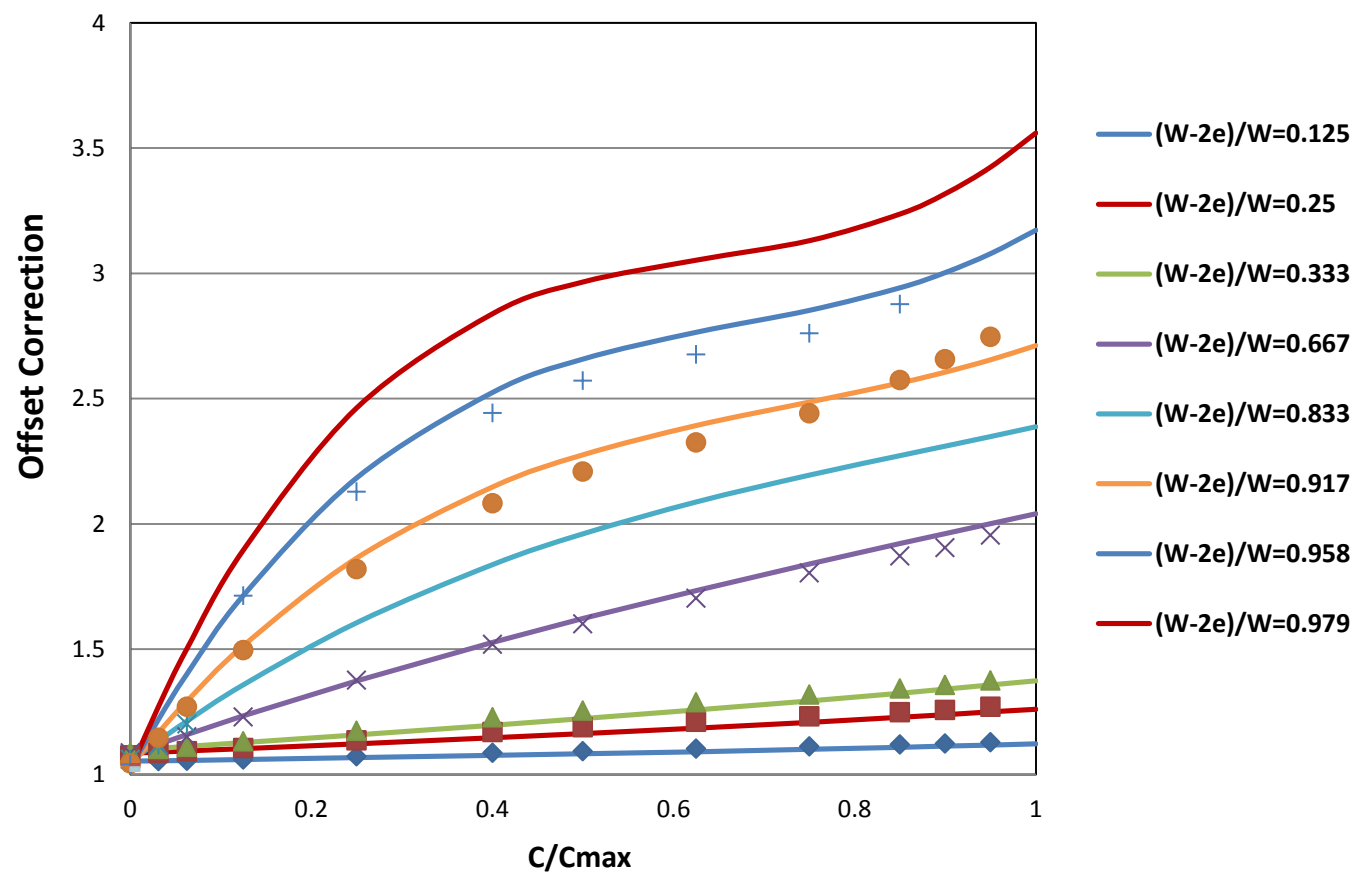
Near Edge Offset Correction

$e/D = 2$



Far Edge Offset Correction

$e/D = 2$



Advanced Model Solution Approach for Small Cracks ($a/t < 0.1$)



- Single Corner at a Hole



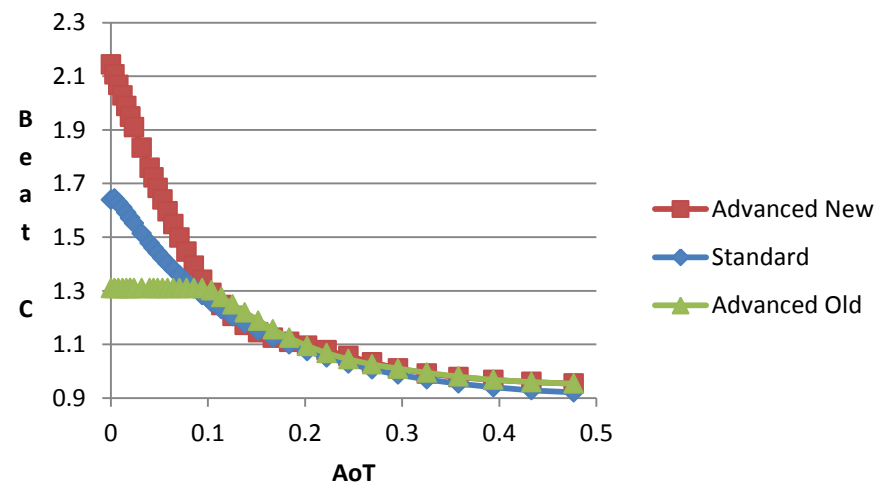
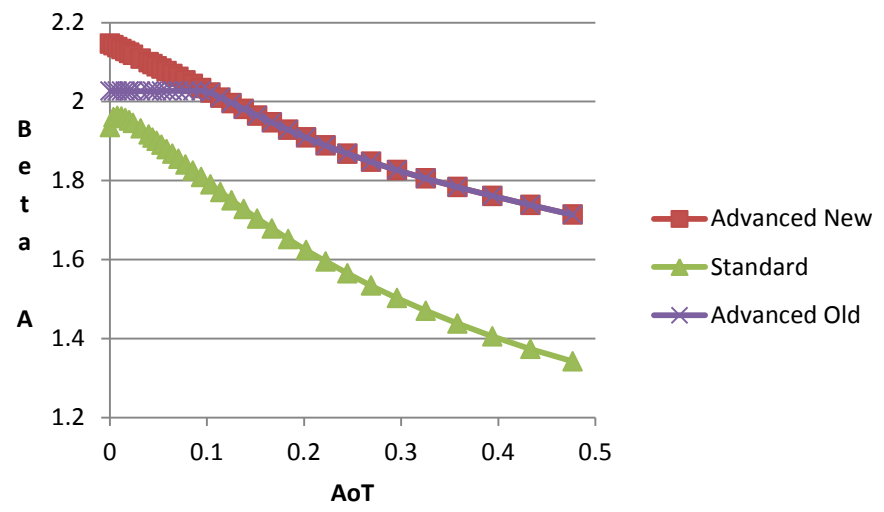
- Double Corner at a Hole

As the crack length goes to zero, the beta for both crack growth directions converge to :

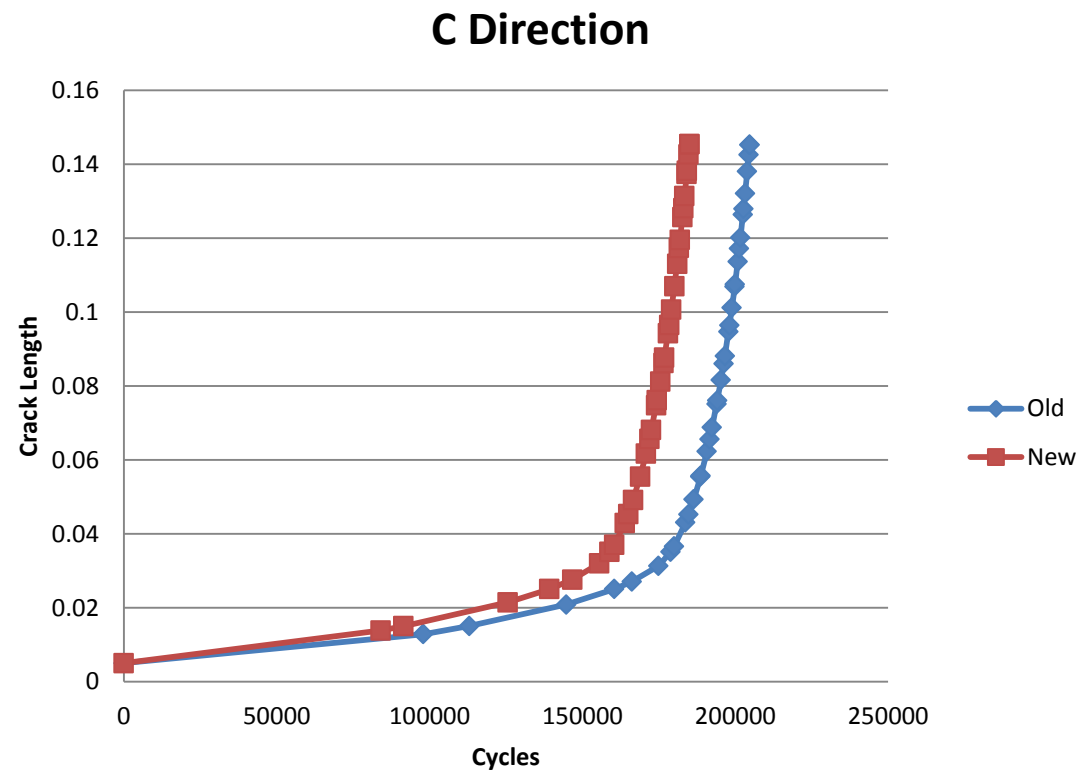
$$\beta = 1.122 K_t \frac{2}{\pi}$$

Tension and Bending = 2.142, Bearing = 0.625 (W/D = 100, same as Fawaz/Andersen)

Advanced Corner Cracked Hole Model (Small Crack Beta Values)



Advanced Corner Cracked Hole Model (Small Crack Life Comparison)



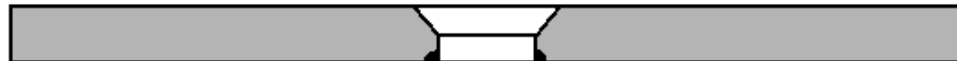
Countersunk Hole Solution

- Base on PHD thesis of the Reinier de Rijck
- Bending, Tension ad Bearing Solutions for the symmetric corner crack

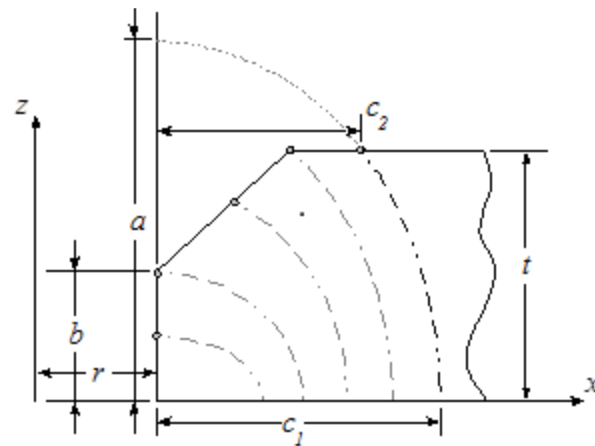
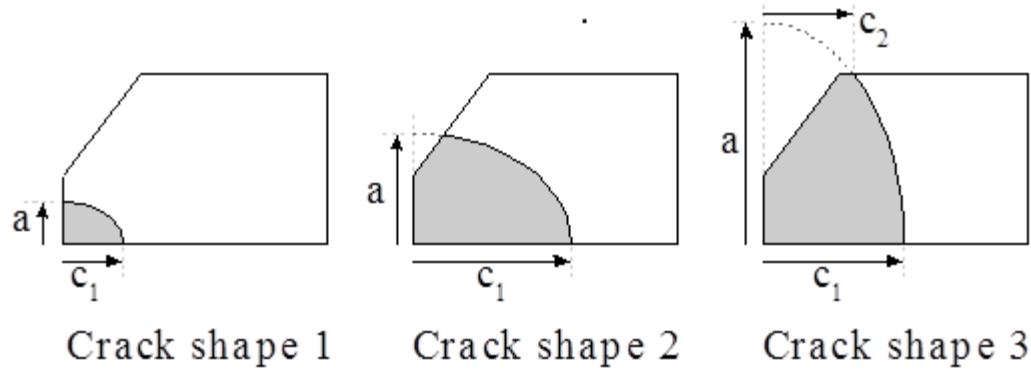
- Centered or offset hole with single corner crack



- Centered hole with symmetric double corner crack



Countersink Dimensions



Countersunk Solution Workspace

Countersink Angle – 100 Degrees

Depth to Thickness Ratios (B/T)

0.05, 0.25, 0.5

Virtual Crack Depth to Thickness Ratios (A/T)

0.025, 0.0294467, 0.0333167, 0.0371867, 0.0410568, 0.0449268, 0.05,
0.149025, 0.24687, 0.344714, 0.442559, 0.540403, 0.646563, 0.752822,
0.849658, 0.946494, 1.04333, 1.14017, 1.25287, 1.47261, 1.6767, 1.8808,
2.0849, 2.28899, 2.50574

Crack Depth to Crack Length Ratios (A/C)

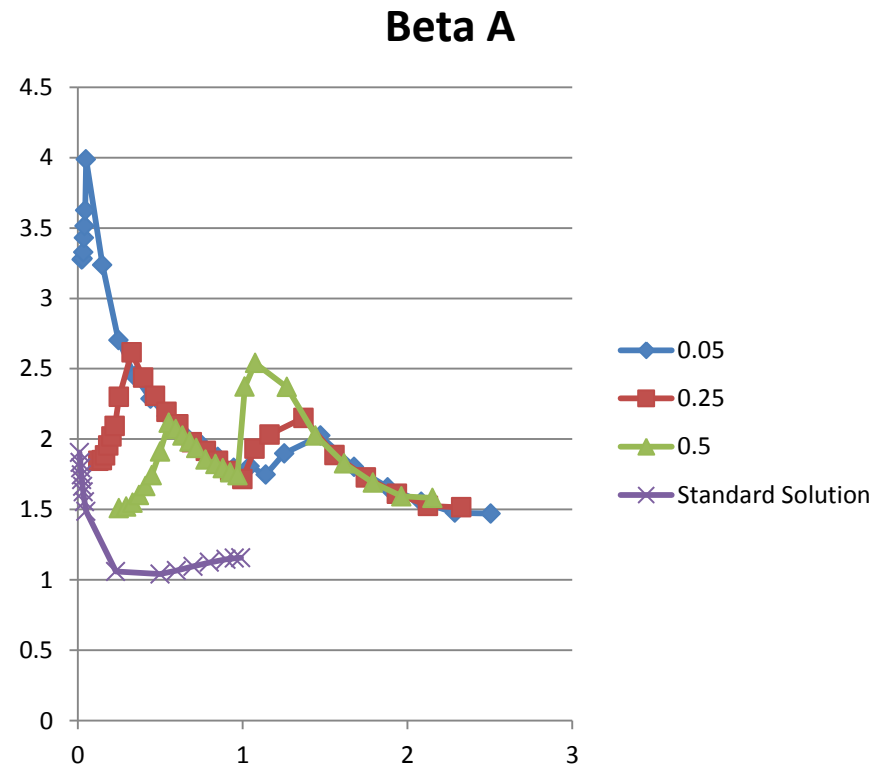
0.5, 0.67, 0.75, 1., 1.5, 2., 5.

Hole Radius to Thickness Ratios (R/T)

0.1, 0.25, 0.5, 0.67, 1.0, 1.2, 1.5, 2.0, 2.4

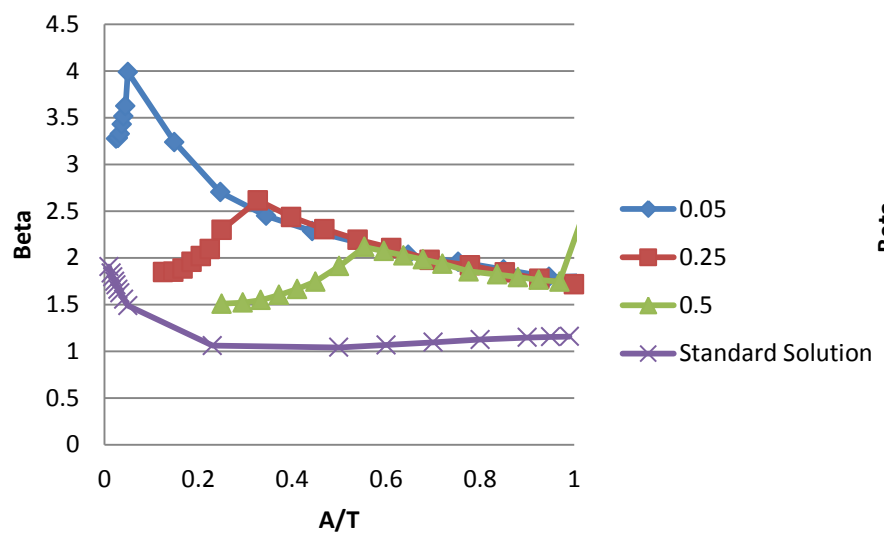
Countersunk Hole Transitions

- Through the knuckle
- To an oblique crack
- To a through crack

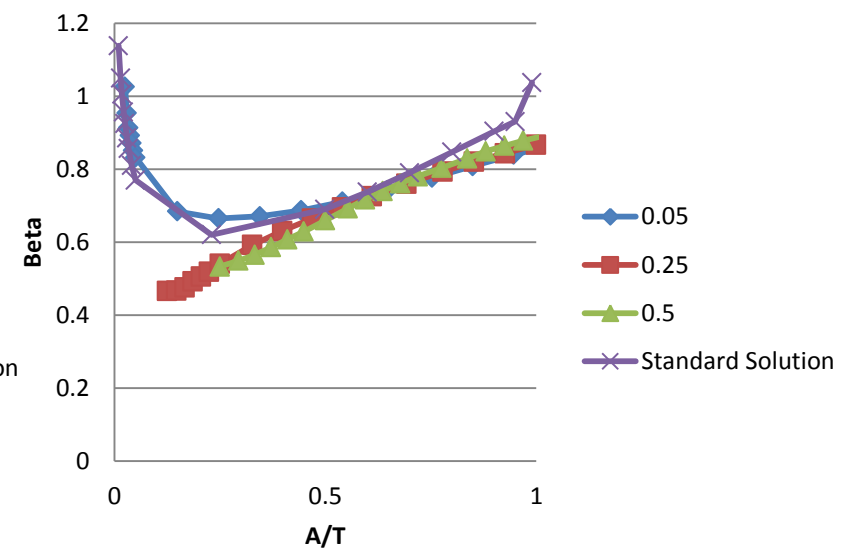


Countersunk Solution Betas

Beta A



Beta C



Spectrum Tool

The screenshot displays the Spectrum Manager application interface, which is divided into several functional areas:

- File Browser (Top):** Contains 'Load / Save' and 'View' tabs. The 'Load / Save' tab includes buttons for 'New', 'New SubSpectrum', 'Open SP3', 'Open XML', 'Save', 'Close', and 'Exit'.
- Data Entry Form (Middle-Left):** A form for entering spectrum details:
 - Spectrum Title:** Falstaff
 - SubSpectrum Label:** Flight
 - Number Of Cycles:** 18619
 - Number Of Blocks:** 16207
 - Maximum Max:** 1
 - Maximum Min:** 0
 - Spectrum Description:** (Empty text area)
 - Save** button
- Sequence List (Middle-Right):** A table with columns 'Sequence' and 'Sub Spectra'.

Sequence	Sub Spectra
Flight 1	Flight 1
Flight 6	Flight 2
Flight 3	Flight 3
Flight 4	Flight 4
Flight 5	Flight 5
Flight 6	Flight 6
Flight 2	Flight 7
Flight 6	Flight 8
Flight 4	Flight 9
Flight 5	Flight 10
Flight 4	Flight 11
Flight 3	Flight 12
Flight 4	Flight 13
Flight 5	
Flight 6	
Flight 7	
- Spectrum Plot (Bottom):** A line graph showing 'Spectrum' on the y-axis (ranging from -0.4 to 1.0) and 'Cycles' on the x-axis (ranging from 0 to 17984). The plot shows a complex, oscillating signal with multiple peaks and troughs.

Crack Closure Factor

- Why do we have it?
- Applied only on a crack at the open edge surface
- Added to standard, user defined, advanced and plug-in k solutions

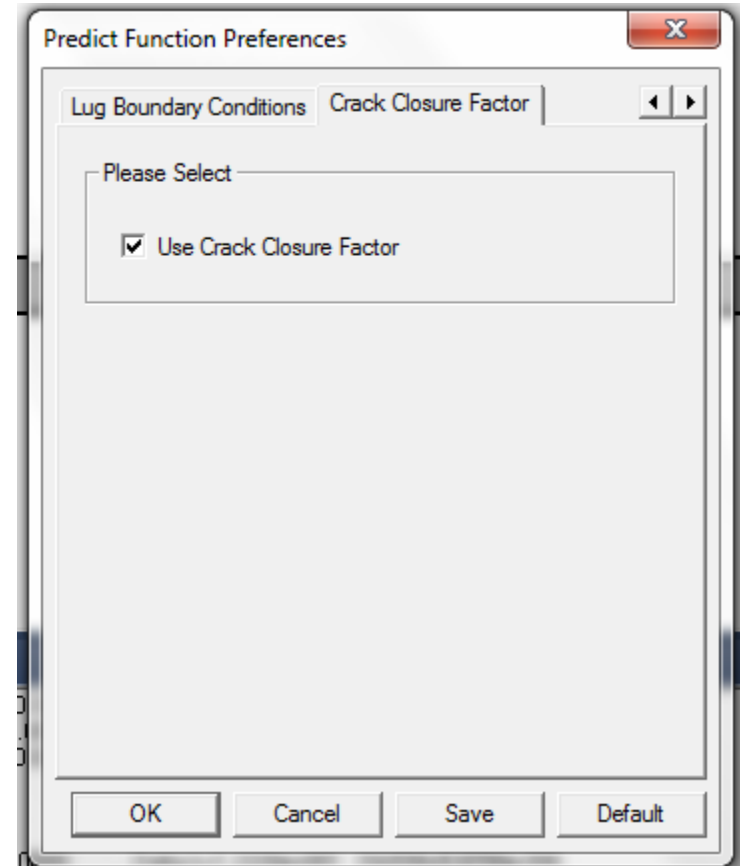
$$K = \beta_r * \sigma \sqrt{Length * \pi} * \beta$$

if $R = 0$
then

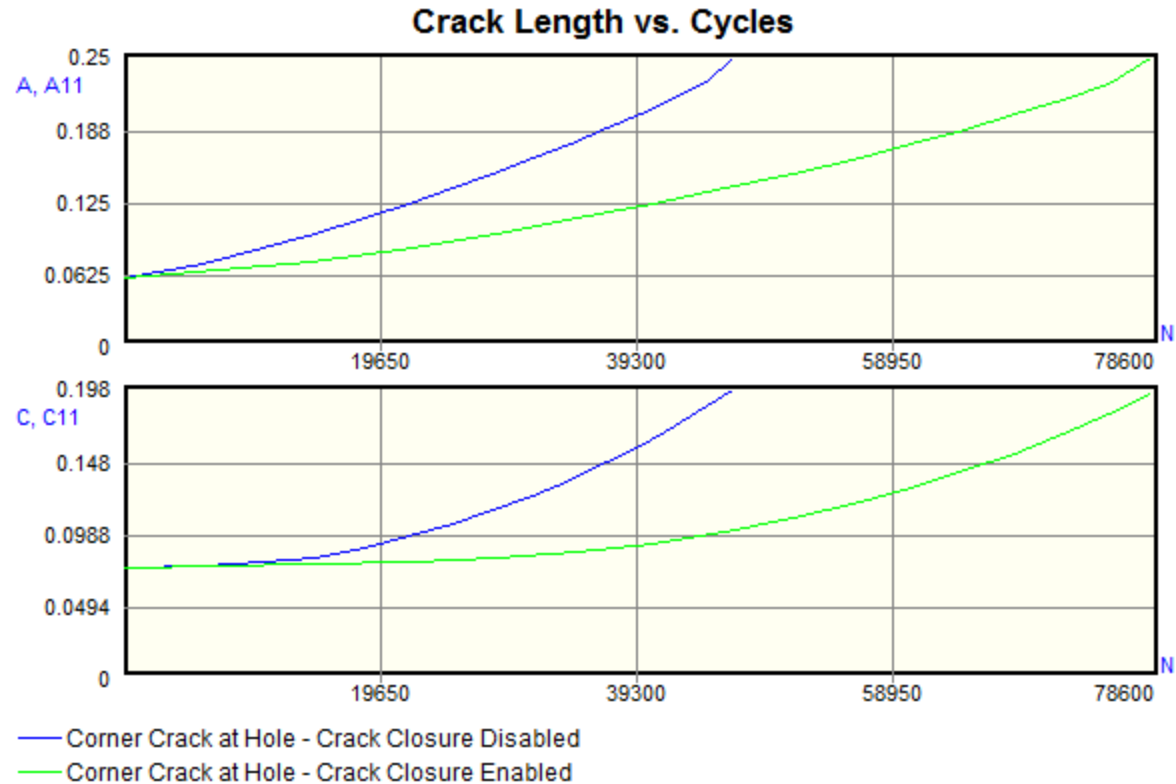
$$\beta_r = 0.9$$

else

$$\beta_r = 0.9 + 0.2 * R^2 - 0.1 R^4$$



Crack Closure Factor Effect



Release 5.3

- New weight functions solutions (crack at hole, stress distribution in C-direction for part through cracks)
- Using different material data as a function of spectrum
- Applying different material data to the different crack directions
- Ability to Replicate Results From Previous Versions (back to Version 5.01)
- Option to Save Input File With Retardation State Data for Later Restart
- Undo-Redo
- New solution for a corner crack at the countersink knuckle

AFGROW Users Workshop 2012

The Annual AFGROW Users Workshop will be held at the Davis Conference Center in Layton, UT on September 8-10, 2012.

Who should attend: Engineers, researchers, technologists, and managers who are users of AFGROW and/or have an interest in Crack Growth Analysis tools.



The purpose of the workshop is to provide a forum for users to share ideas, demonstrate how AFGROW is being used for practical cases, and discuss future development. We will also provide information on the latest development efforts.

Questions